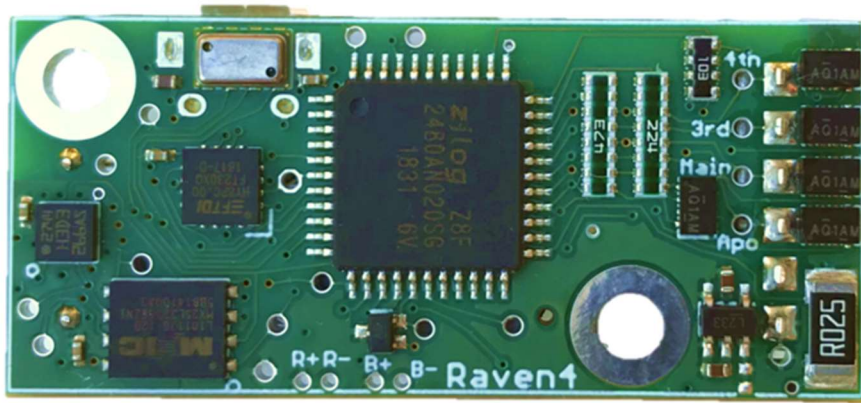


# Raven User's Manual

Updated for the Raven4 altimeter  
August 15, 2020

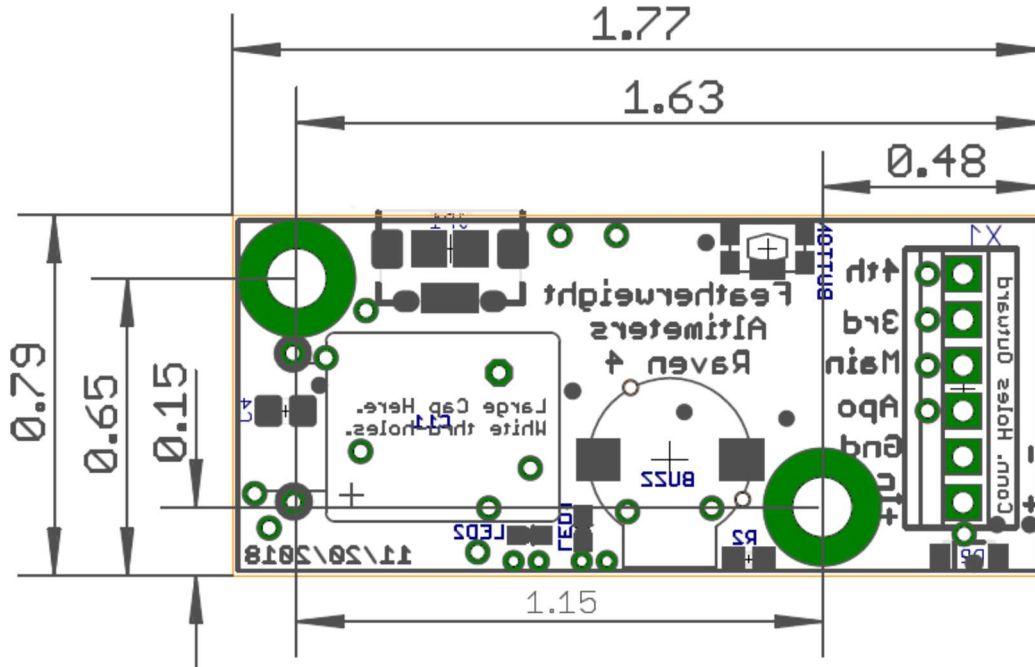


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## Mounting and Installation:

The Raven is mounted with its long axis parallel to the direction of flight. Either end can be up; the altimeter senses the orientation while on the pad and will work either way. The mounting holes accommodate #4 size screws. Two #4-40 screws, two spacers and two #4-40 nuts are provided. The dimensions of the board are shown below, in inches:



## Deployment Connections:

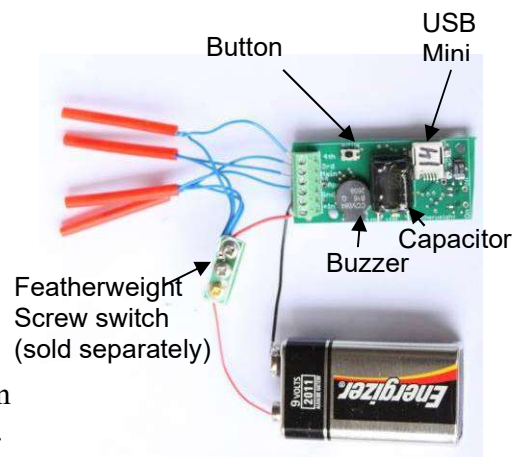
The following diagram and picture show how to connect the Raven for deployments, using a single switch that turns on the unit and arms the deployment charges:

A description of the default program for the 4 outputs is given in the [deployment programming section](#) of this document.

### Caution:

- Do not use this or any other altimeter for deployments until you have performed a ground test to verify that the Raven will work correctly in your rocket and with your pre-launch procedures. (See "Running a flight simulation" section)

Flying the Raven in a rocket with motor ejection or another altimeter you are familiar with is recommended for your first Raven flights. Each Raven has passed a basic functional test before shipment, but you alone are responsible for



ensuring that your altimeter works for your application and that your rocket will fly safely.

### ***Power:***

An external battery and a switch are required. The Raven can be powered by any DC power source between 3.8 and 16 V. 9V batteries and single LiPoly cells of 165 mAh or less are the most common choices. An aerogel ultracapacitor will keep the altimeter operational when the power is disconnected for 7-10 seconds, so a separate deployment battery is not required. If one is desired, a pyro battery up to 20V can be connected through the deployment charges while a lower-voltage battery is used for the electronics via the +In and Gnd pins. Please carefully note the battery polarity before connecting it. Although the Raven4 has reverse battery protection, connecting a large battery in reverse could permanently damage the altimeter.

The field-effect transistor (FET) used to switch all four deployment outputs is rated for 22 Amps for pulses  $\leq 1$  second, and has a resistance of about 11 mOhms. The largest lithium polymer battery that should be used with the Raven is 1S, 300 mAh, because larger lipo batteries can produce currents well in excess of the 22 Amp rating of the deployment channels in the event of an output short circuit, which can permanently damage the altimeter. Motor airstarts typically result in short circuits as soon as the motor ignites, so be sure to use batteries for airstarts that cannot exceed the FET rated current when shorted.

Any external switch can be used as long as it can handle the deployment charge current. Our customers love the Featherweight screw switch because of its reliability, tiny size, and convenient mounting terminals. The screw head wipes and preloads the contacts with every operation.

### ***High G flights:***

For extra durability, a dab of epoxy or silicone can be used under the Raven's hold-up capacitor to secure it to the board. The hold-up capacitor is the relatively component wrapped in shiny blue or black plastic next to the USB connector.

## **Operation in the Field**

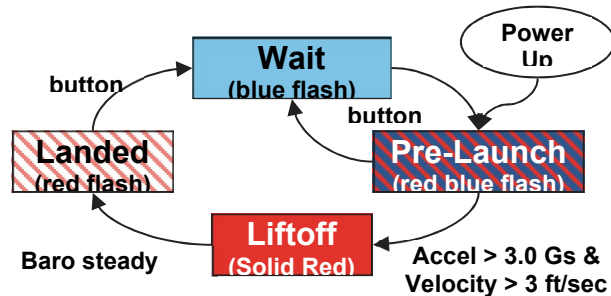
Although the Raven is designed as an advanced recording altimeter, it can be installed and used indefinitely without ever being connected to a computer, using the default settings for the outputs. The following describes how the Raven is operated at the launch range.

### ***Pre-flight operation***

Turn on the Raven using an external arm switch. The first set of beeps after power-up is the battery voltage, rounded down to the nearest volt. (9.7 Volts = 9 beeps). After power-on, the Raven goes into pre-launch mode. The Raven will beep a low, single beep every 2 seconds if no charges are detected or if the accelerometer does not read a near-vertical orientation. In the case of the Raven3 and Raven4, the single beep will also happen if the battery voltage is below 3.85V. The Raven beeps once for each channel and indicates the continuity for each channel by giving a high beep for channels with high continuity voltage, and a single beep for channels without a charge connected. The

blue LED flashes once per high beep and the red LED flashes once per low beep. The order of the channel beeps is the same as the order of the channel outputs along the terminal block. First is the Apogee channel, then main, then 3<sup>rd</sup>, then 4<sup>th</sup>. In prelaunch mode, the blue LED will flash once per second, and the red light will flash during beeps. The button, located on the terminal block side of the board, can be used to switch between wait and prelaunch modes.

The diagram (right) shows the different operating modes of the Raven and the transitions from one mode to another.



**Caution: If electronic deployments are expected, do not launch unless the Raven is beeping with the expected number of connected output charges.**

### ***Liftoff Detection and flight***

In prelaunch mode, the Raven is watching the axial accelerometer to detect liftoff. Any readings less than 3.0 Gs are ignored. The Raven is also updating the accelerometer calibration offset it uses during the flight, to maximize the accuracy of its apogee detection. If the rocket is dropped a short distance, the Raven will ignore the subsequent contact to avoid premature liftoff detection.

**Despite these features, it is possible for the Raven to misinterpret normal handling as a liftoff, so do not arm and/or turn on the Raven with charges connected until the rocket is installed in the launch rail or tower and the rocket is pointed in a safe direction.**

The Raven will detect liftoff when accelerometer readings in excess of 3 Gs add up to a 3 feet/second upward velocity.

In liftoff mode, the Raven's red LED lights continuously, and the following data is stored:

#### Periodic measurements:

- 400 Hz axial Accelerometer, +/- 105 Gs
- 200 Hz lateral Accelerometer, +/- 105 Gs
- 20 Hz Baro data, +/- 0.3% accuracy
- 20 Hz voltage on each of 4 outputs
- 40 Hz output current
- 20 Hz high-precision temperature sensor
- 20 Hz for all flight events used for deployment logic.

#### Once per flight:

- Flight counter
- All output program settings
- Accel calibrations used during the flight

- Pad altitude ASL

After the first 6 minutes of flight, the data recording changes to a lower rate, and stores approximately 45 minutes more. The liftoff mode will continue, and deployment outputs will function if necessary, after the end of the data recording. The liftoff mode only ends when landing is detected.

When the flight conditions assigned for each output are true, ([See configuring the Raven using the FIP section](#)) the output switch will turn on. The default deployment conditions are the following:

<b>Output function</b>	<b>Apogee deployment</b>	<b>Main chute deployment</b>	<b>Backup apogee deployment</b>	<b>Backup Main deployment</b>
Altimeter output Label	Apo	Main	3 <sup>rd</sup>	4th
Upward velocity (from accel)	< 400 ft/sec	< 400 ft/sec	< 0 ft/sec	< 400 ft/sec
Baro altitude	Decreasing	Decreasing	Increasing or decreasing	Decreasing
Altitude above pad	Any	< 700 feet	Any	< 700 feet
Time delay	0	0	2 sec	1.5 sec
Burnout Counter	1	1	1	1

These default output settings can easily be changed using the Featherweight Interface Program (FIP).

### ***Post-flight operation***

When the Raven is below 2000 feet above the launch pad, it watches for the barometric readings to become constant to detect landing. After landing, the altitude at apogee will be beeped out once and then the red LED will flash once per second. These features save battery life.

To hear the peak altitude after recovery, pick up the rocket or av-bay and tilt it one direction and then another. When the Raven detects that it has been flipped over, it will beep out the altitude again. The Raven will beep out each decimal of a 5 decimal altitude, in feet, starting with the 10,000 place. A pause between sets of beeps indicates a change to the next decimal place. A short, low beep indicates a 0 in that decimal place. For example, 1024 feet is represented as <low beep, for zero in the 10,000 place>, pause, <one high beep> pause <one low beep> pause <2 high beeps> pause <4 high beeps>. The button can be used to switch between post-landing mode and wait mode. Whenever the mode is switched into wait mode, the last apogee altitude will be repeated.

## **Computer-Attached Operations**

### ***Featherweight Interface Program (FIP) Installation:***

Download the FIP installation package from the Featherweight site. The installation package contains both the FIP and the USB drivers necessary for the Raven to interface with the computer. Double-click on the zip package and follow the installation package instructions.

## Connecting to the Raven using the FIP:

The Raven4 uses a USB-micro connector, so a USB-micro cable typically used in cell phones, digital cameras, etc. will be compatible. Start up the FIP. If the Raven altimeter is already plugged in, go to the “Altimeter” menu and select “Connect.” If the Raven altimeter is not yet plugged in, plug it in now and the FIP will detect the connection to the Raven. The following dialog box will appear:

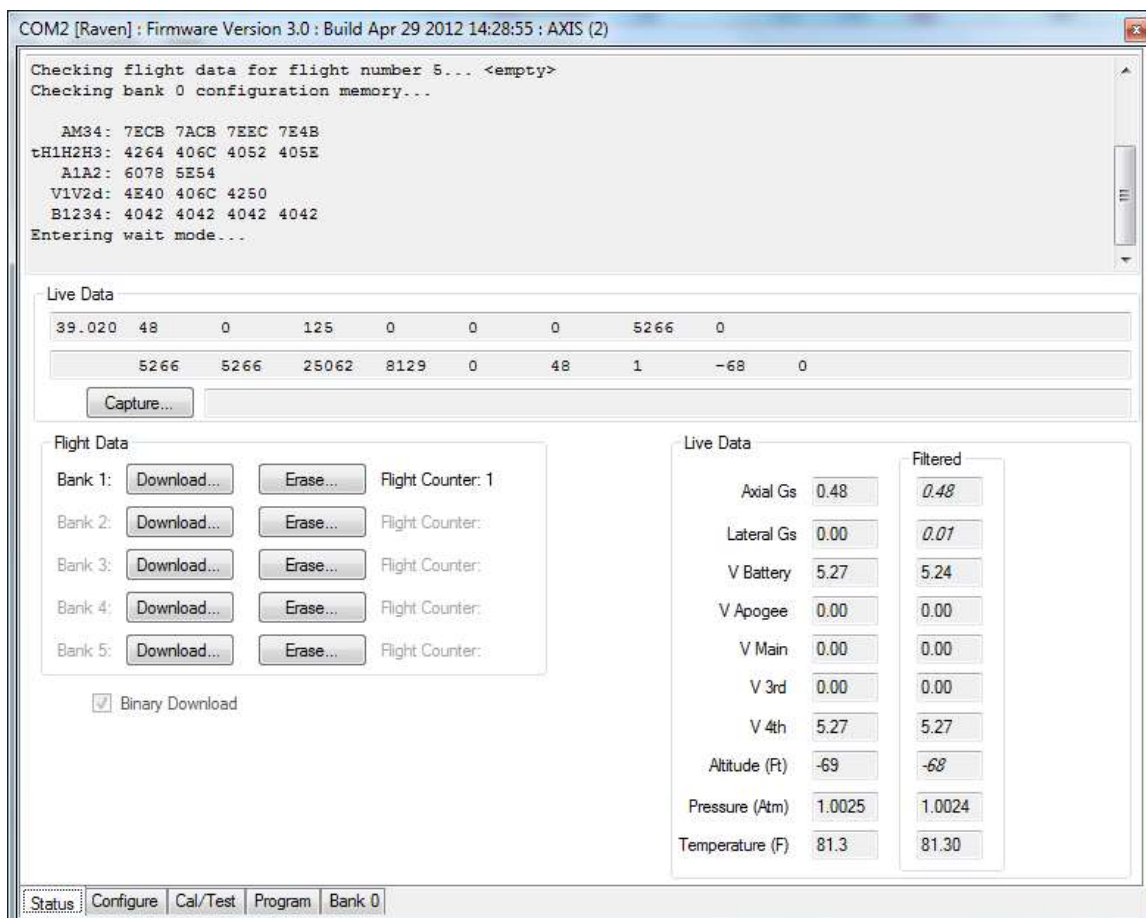


Select the correct altimeter model and the comm port to which the Raven is connected. If multiple comm ports show up in the dialog box, the highest-numbered one will probably be the one with the Raven connected to it.

The FIP will check to see which of the 5 flight indexes have data in them, and will begin to provide real-time information from the Raven, including barometric pressure (in atmospheres), accelerometer readings, and temperature.

## Raven Status and Data Download

The status tab of the FIP shows the current status of the Raven. The Flight Data box shows the status of each of the Raven’s 5 memory banks, and which flight corresponds to each data location.



```
Checking flight data for flight number 5... <empty>
Checking bank 0 configuration memory...

AM34: 7ECB 7ACB 7EEC 7E4B
tH1H2H3: 4264 406C 4052 405E
A1A2: 6078 5E54
V1V2d: 4E40 406C 4250
B1234: 4042 4042 4042 4042
Entering wait mode...
```

Live Data									
39.020	48	0	125	0	0	0	5266	0	
5266	5266	25062	8129	0	48	1	-68	0	

Flight Data		
Bank 1:	Download...	Erase... Flight Counter: 1
Bank 2:	Download...	Erase... Flight Counter:
Bank 3:	Download...	Erase... Flight Counter:
Bank 4:	Download...	Erase... Flight Counter:
Bank 5:	Download...	Erase... Flight Counter:

Binary Download

Live Data		
Axial Gs	0.48	Filtered 0.48
Lateral Gs	0.00	0.01
V Battery	5.27	5.24
V Apogee	0.00	0.00
V Main	0.00	0.00
V 3rd	0.00	0.00
V 4th	5.27	5.27
Altitude (Ft)	-69	-68
Pressure (Atm)	1.0025	1.0024
Temperature (F)	81.3	81.30

Status | Configure | Cal/Test | Program | Bank 0

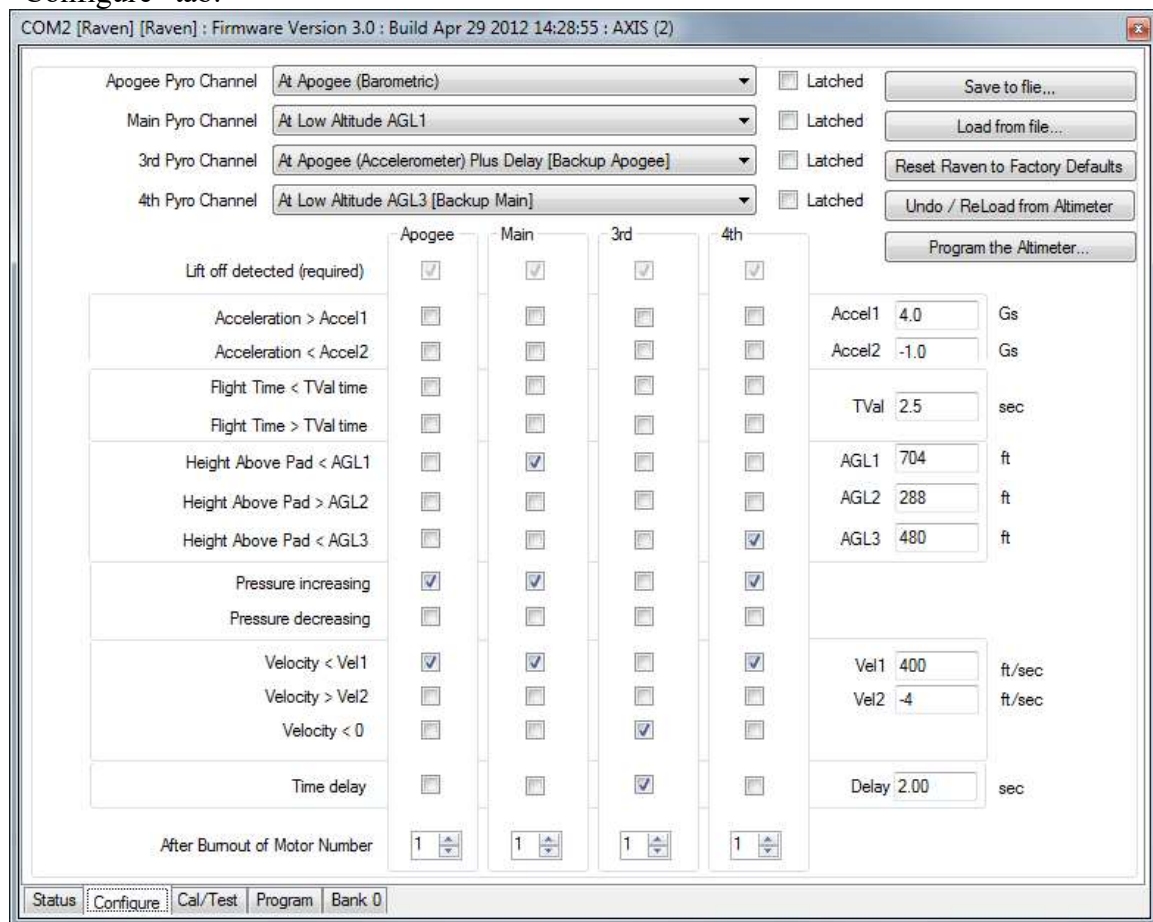


In the example above, 1 out of the 5 flight banks are full. The most recent flight (#1) is located in bank 1.

The live data box provides real-time status and measurements from the Raven. To download data, select the button “Download” for the flight data you’re interested in, and follow the dialog box instructions to save the data to a file. The FIP automatically detects the end of the flight, but you can interrupt the download before the end of the flight, if desired. After the download is complete, the FIP switches to the data viewing window. A data file from a previous download can be opened for viewing using the File/Open command.

## Configuring the Raven using the FIP:

To program and verify the deployment options programmed into your Raven, click the “Configure” tab.



The logic used for each deployment output is represented as a column of check boxes in the dialog box. The headings “Apogee”, “Main”, “3<sup>rd</sup>” and “4<sup>th</sup>” correspond to the text labels on the altimeter. All of the channels can be programmed to be used for any purpose. The text boxes arranged along the right side are set points that you can change, that control the speed, altitude, etc. the altimeter will check for when deciding when to fire the output. A deployment channel is triggered when all of the checked flight events are true. Note that most flight events, like acceleration < 0, can switch from true to false

and back again throughout the flight. Other events, like  $\text{time} > \text{user timer value}$ , can have no more than one transition during the flight.

The “time delay” option will delay the start of the output by the specified time delay after all other conditions are met. Note that once the deployment conditions are met, the deployment will happen after the time delay regardless of any changes in the deployment conditions after the delay timer has started. Each output will be activated for 1 second, unless the “Hold the switch closed continuously” option is selected for that output. The continuous output option is designed to turn on a non-deployment load, like a transmitter. This option is not recommended for use with deployment charges because charges can have residual shorts that could drain the pyro battery for future use and/or damage the FET switch.

## Deployment triggers details and examples:

### **Pressure Increasing (used for the default apogee, main, and main backup deployment channels):**

This condition is true whenever the filtered baro pressure measurements are increasing. Since pressure decreases with altitude, this trigger corresponds to when the rocket is falling. This trigger is used for the default main deployment channel so that the ascent is ignored for the main chute deployment. Transitions to and from supersonic flight can cause this condition to become true temporarily during ascent. That’s why the main deployment channel looks for an additional condition, the velocity check.

### **Velocity < 0 feet/second (used for the backup apogee deployment channel)**

This trigger is true when the upward velocity has stopped and the rocket is starting to fall. The Raven senses the acceleration and subtracts off what it was reading when it was sitting on the pad (about 1 G). Then it adds up the accelerations continuously throughout the flight to calculate the velocity. This method provides an apogee detection that is immune to pressure anomalies, but off-vertical flights or mis-aligned altimeter can affect the apogee detection accuracy.

### **Height above pad < AGL1: (used for the default main deployment channel)**

This condition is true when the measured altitude above the pad is below the AGL1 setting. AGL1 can be set between 0 and 32736 feet, in increments of 32 feet. For the rest of the main deployment settings, other conditions need to be checked, since this condition will be true from liftoff until the rocket gets above the main deployment altitude.

### **Velocity < Vel1: (used for the default main deployment channel and backup channels):**

This condition is true when the accelerometer-based estimate of upward velocity is lower than Vel1. This trigger is useful in combination with the pressure increasing trigger, to ensure that the rocket is out of the transonic region before using the pressure increasing check. The velocity check is also a useful trigger for 2<sup>nd</sup> stage ignition. Optimal altitude can be achieved by allowing the rocket to slow down before 2<sup>nd</sup> stage ignition, while maintaining enough speed so that the flight path stays nearly vertical. This trigger can be used to ignite a 2<sup>nd</sup> stage at the desired velocity, even in the event of unexpected drag or motor performance.



**Velocity > Vel2:**

This condition is true when the accelerometer-based estimate of upward velocity is higher than the Vel2 value. Vel2 can be set between -48 feet/second and 1998 feet/second, in 2 ft/second increments. This setting is one way to check to make sure that the rocket is flying as expected before performing a high-speed sustainer ignition. If the rocket is in the middle of a tumbling breakup, the velocity will be lower than expected

**Acceleration > Accel1 and Acceleration < Accel2:**

This condition is true when the accelerometer's reading is greater than the Acc1 value or less than the Acc2 value, respectively.

Acc1 and Acc2 can be set in the range from -50Gs to +50Gs, in increments of 0.1 Gs. These triggers are useful for detecting staging charges, deployment charges, motor burns, and landing events. Note that there are lots of different flight events that can cause changes in the acceleration values, so it's important to combine this trigger with other conditions if used. For example, you could use this trigger to detect landing for a chute disconnect device, by looking for acceleration greater than 3 Gs. For this application, you would also need to prevent a premature deployment by setting the following additional conditions for the trigger: the altitude < AGL1, pressure increasing, velocity < Vel1 to avoid a premature deployment. New for the Raven2 (not available with the original Raven) is a third altitude trigger that makes it easier to set up the landing detection and the main deployment using the same altimeter (See Height Above Pad < AGL3)

**Time < user timer value and Time > user timer value:**

This condition is true when the elapsed time from liftoff detection is less than, or greater than, the user-settable timer threshold, TVal. TVal can be set from 0 to 51.2 seconds, in 0.02 second increments. The time > Tval setting can be used as a simple timer, or in combination with other conditions to make sure that the rocket has gotten past the initial low-altitude part of the flight. Note that liftoff detection may occur a fraction of a second after 1<sup>st</sup> motion.

**Height Above Pad > AGL2**

This condition is true when the height above the pad is greater than the user-settable altitude AGL2. AGL2 can be set between 0 and 32736 feet, in increments of 32 feet. This trigger is useful for air start ignition of motors, as it can be combed with the time < TVal to ensure that the rocket is on its way to a nominal flight before igniting the next motor. For example, if a simulation predicts that the booster + sustainer should achieve 7000 feet within 6 seconds, you could set the AGL2 to 6000 feet and the Tval to 6 seconds to be assured that the sustainer won't fire if the rocket is tumbling or has a flight angle far from vertical. To ensure that the measured pressure won't be affected by Mach transition transients, you could also check to make sure that the pressure is decreasing and the velocity is < 1000 ft/second.

**Height Above Pad < AGL3 (used for the default main backup channel)**

This condition is true when the height above the pad is greater than the user-settable altitude AGL3. AGL3 can be set between 0 and 32736 feet, in increments of 32 feet. This trigger is useful for triggering an event below the main altitude. For example, the main deployment altitude could be set to 1000 feet, and a backup main deployment at 800

feet. This trigger is also useful as part of a landing detection. To detect landing, set AGL3 well below the main deployment altitude but higher than the highest terrain that the rocket might land on. Set the velocity  $< 0$  check (so that it won't happen on the way up), and set the accelerometer check consistent with the direction of the landing impulse. For a typical setup with the rocket's main chute attached above the av-bay, look for a positive acceleration, and use a value that is higher than the rocket might get on the main chute ( $> 1.5\text{-}2$  Gs), but lower than the landing impulse (typically  $5\text{-}20$  Gs). Note that the default settings in the FIP for landing detection assume that the av-bay will land in the same orientation as it has on the pad. If your rocket configuration leads to a nose-down landing, the landing detection should be set up to look for  $\text{accel} < \text{Accel2}$ , and  $\text{Accel2}$  should be set to a negative number.

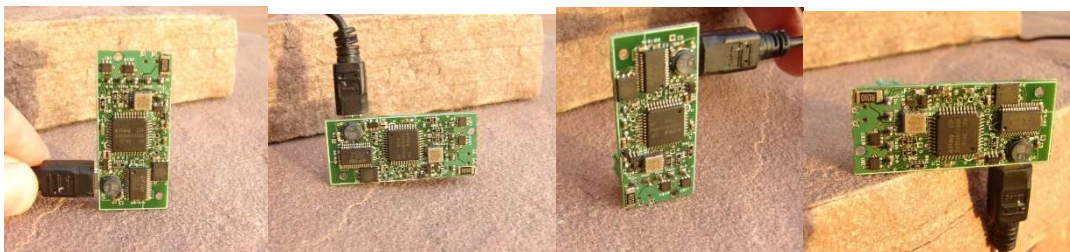
#### **Burnout counter (default of 1 for all outputs):**

This condition is the count of the motor burnouts detected by the Raven. It is settable between 0 (don't wait for any burnout indication) and 1023 (not that anyone can make a 1024 stage rocket!), and the burnout count can be set individually for each output. The Raven watches for a change in velocity of at least 40 ft/sec, and then waits for the velocity to drop by at least 5 feet/second before adding 1 to the burnout count. When the burnout counter is greater than or equal to the user burnout count for that output channel, the trigger will be true.

Unlike the burnout counter used in other altimeters, this version will not mistakenly trip if a launch rod snag or noisy motor burn causes a brief drop in the motor thrust. It also won't mistake a stage separation charge from a motor ignition. However, very low-impulse airstart motor may not trigger the burnout counter, so this should only be used for motors that will increase the velocity by 40 feet/second or greater.

#### **Calibrating the accelerometer:**

The Raven has a user-calibration function for the accelerometer to ensure correct accelerometer operation and account for accelerometer drift over time. If the average axial G reading in the live data is outside of 0.8 to 1.2, the accelerometer should be recalibrated. To calibrate the accelerometer, push the "calibrate" button in the cal/test flight tab and place the Raven on a flat surface on its short edge. Follow the FIP instructions to hold the Raven in each of the 4 positions shown. Each position is a 90 degree rotation from the previous position.



#### **Running a flight simulation**

The Raven has a flight simulation feature that allows a realistic flight simulation to be performed on the Raven, using the Raven's own sensors, the current deployment logic, and real activation of the outputs. This is useful for verifying the deployment settings, as well as the compatibility of the Raven power source with the e-matches or igniters used.

The flight simulation is available on the cal/test tab of the FIP. Follow the on-screen directions and button labels to perform the flight. You can control the duration of the 5G acceleration, even turning it on and off to simulate multiple-stage flights or air-starts. The Raven records all the data just as if it were a real flight, so you can familiarize yourself with the Raven operation and data review.

The simulated flight assumes that the Apo output channel is used for apogee deployment, and the Main output channel is used for main deployment. When the apogee channel fires, the simulation applies a 100 ft/second descent rate. When the main output fires, a descent rate of 20 feet/second is applied. The other two outputs have no effect on the simulated flight trajectory.

Although the Raven can be mounted in your rocket with either end up, the simulated flight must be conducted with the screw terminal block up. This is because the simulated thrust is implemented as a test mode of the accelerometer chip that applies the acceleration in a single direction.

**Caution: Do not run a flight simulation with e-matches or igniters connected unless the igniters/e-matches are located away from flammable materials and adequate ventilation is provided. In particular, do not run a flight simulation with black powder charges or motors connected!**

### ***Viewing the data***

A data file from a previous download can be opened for viewing using the File/Open command. The default set of measurements shows the altitude above the pad in feet, and the axial accelerometer trace. Additional traces can be added to the plot using the Parameter selection tool bar. Multiple measurements can be plotted on the same time scale by holding down the <control> key when selecting the traces.

The lower section of the parameter selection toolbar allows you to select flight event logic that was recorded during the flight. The flight event logic states are the conditions that are measured by the Raven and used in the deployment control logic. The deployment output fires when all of the checked logic conditions are true. By plotting the voltage on each of the outputs and the flight event logic that was checked during the flight, you can verify exactly what the altimeter was firing, when, and why. This is useful for verifying a flight program by looking at the data recorded in a simulated flight, which records data just as a real flight would.

To zoom in on a particular range of time, drag with the left mouse button just underneath the time axis. Click the right mouse button to zoom out or type in the axis ranges. To zoom in on the vertical axis for a measurement, drag with the left mouse button on the measurement axis you want to change.

To export formatted data into a separate analysis tool (such as Matlab or a spreadsheet), right-click in the parameter selection window, which brings up options for saving as a delimited file or copying data to the clipboard, where it can be pasted directly into a spreadsheet.

## Calibration and Accuracy:

The barometric sensor used in the Raven is a digital-output sensor that contains the entire analog measurement chain on one chip. The sensor is factory-calibrated over a wide range of temperatures and pressures to provide exceptional accuracy under any rocketry conditions. The Raven has less than 1 mbar (0.1% full-scale) pressure error over most conditions, and less than 3 mbar pressure error (0.3%) the temperature range from -40C to 85C, and from 1.1 Atm (2500 feet below sea level) to 0.3 Atm (30,000 feet above sea level).

Keep in mind that however accurately the Raven measures pressure, converting pressure to altitude results in additional error. The Raven uses the International Standard Atmosphere (ISA) model, which uses 3 different formulas for different altitude ranges between sea level and 104,987 feet to compensate for the temperature behavior of different parts of the atmosphere. The standard atmosphere model is implemented in the Raven with full ANSI-C floating point calculations to avoid errors from numerical approximations. The standard atmosphere model, however, is an approximation to typical conditions for mid-latitude locations. It assumes a temperature profile that is likely to be colder than typical rocketry conditions. For example, the standard atmosphere model assumes that the sea-level temperature is 59 F, and that the temperature at 5400 feet ASL is 40F. Errors caused by the atmosphere being warmer than the standard atmosphere can result in reported altitudes that are low by 10% or more. For the most accurate conversion between the pressure and altitude, use the twice-daily balloon sounding data measured by NOAA and conveniently available at <http://weather.uwyo.edu/upperair/sounding.html>. Future versions of the FIP may include an option for automated lookup and correction based on this data.

## Specifications:

Brand Model	Featherweight Raven4
Axial Accel range and frequency	105Gs, 400Hz
Axial Accel resolution	0.045 Gs
Lateral Accel range and frequency	105 Gs, 200 Hz
Lateral Accel Resolution	0.09 Gs
Download Interface	USB micro
Baro Range (kft)	100
Baro resolution	0.00004 atm
Pyro Outputs	4
Max output Amps	22
Other recorded measurements	Temperature, 4 continuity voltages, all event logic, battery V
High-rate recording time	480 seconds
Low-rate recording time	44.8 minutes
Size	0.79" x 1.77" x 0.5"
Mass	6.6 grams

## **Contact Information:**

Email: [featherweightaltimeters@gmail.com](mailto:featherweightaltimeters@gmail.com)